



# CHAPTER 5 Response and Stability Analysis in Frequency Domain

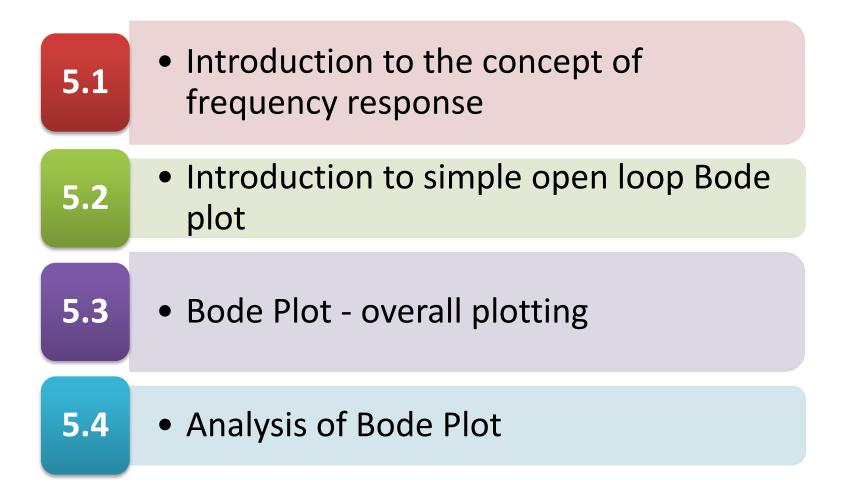
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# **Chapter Outline**







# 5.1 Introduction to the concept of frequency response





# The Concept

- Sinusoidal inputs to a linear system generate sinusoidal responses of the same frequency.
- However, they differ in amplitude and phase angle from the input.
- Sinusoids can be represented as complex numbers called *phasor*.

$$M_1 \cos(\omega t + \phi_1) = M_1 \angle \phi_1$$

where  $M_1$  and  $\phi_1$  are the amplitude and phase angle.

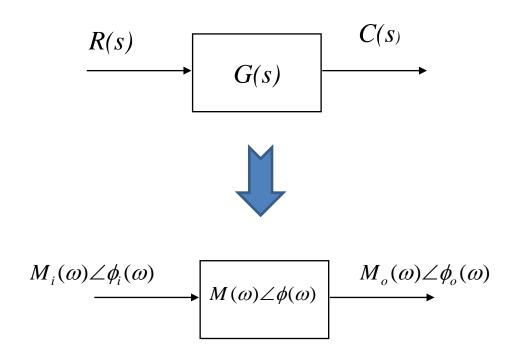
• Thus, the system can also be represented by a complex number so that the product of the input phasor and the system yield the phasor of the output.





# The Concept [2]

• The block diagram:



• The output sinusoid is found by multiplying the input and the system.





# The Concept [3]

Thus  $M_o(\omega) \angle \phi_o(\omega) = M_i(\omega) M(\omega) \angle [\phi_i(\omega) + \phi(\omega)]$ 

Magnitude frequency  $M(W) = \frac{M_o(W)}{M_i(W)}$ response

Phase frequency response

$$\phi(\omega) = \phi_o(\omega) - \phi_i(\omega)$$

• The combination of the magnitude and phase frequency responses is called the frequency response,

 $M(\omega) \angle \phi(\omega)$ 

• The frequency response of a system with transfer function G(s) is  $G(j\omega) = G(s)|_{s \to j\omega}$ 



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# **Plotting Frequency Response**

• For a given transfer function: G(s) = 1/(s+2)

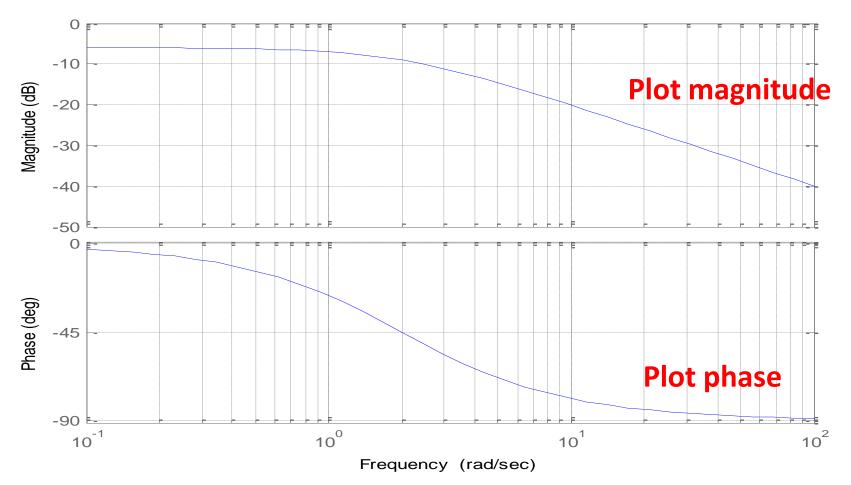
$$G(j\omega) = \frac{1}{(j\omega+2)}$$
$$|G(j\omega)| = \frac{1}{\sqrt{(\omega^2+4)}}; \quad \phi(\omega) = -\tan^{-1}(\omega/2)$$

- One way to plot the Frequency response is by using a separate magnitude and phase plots.
  - a) Magnitude curve: decibel (dB) vs log ω [dB = 20 log
     M]
  - b) Phase curve: phase angle vs log  $\omega$





• Separate plots (magnitude and phase vs log  $\omega$ ).







# 5.2 Introduction to simple open loop Bode plot





# **Bode Plots**

- The log-magnitude and phase frequency response curves as a function of log  $\omega$  is called Bode plots.
- Bode plot is a technique for analyses and design of control systems.
- Consider a transfer function

$$G(s) = \frac{K(s+z_1)(s+z_2)\cdots(s+z_k)}{s^m(s+p_1)(s+p_2)\cdots(s+p_n)}$$

• The magnitude frequency response

$$|G(j\omega)| = \frac{K|(s+z_1)||(s+z_2)|\cdots|(s+z_k)|}{|s^m||(s+p_1)||(s+p_2)|\cdots|(s+p_n)||}_{s \to j\omega}$$

• Converting into dB

 $20\log|G(j\omega)| = 20\log K + 20\log|(j\omega + z_1)| + 20\log|(j\omega + z_2)| + \cdots$  $-20\log|(j\omega)^m| - 20\log|(j\omega + p_1)| - 20\log|(j\omega + p_2)| - \cdots$ 





## **Bode Plots**

• The phase frequency response

 $\angle G(j\omega) = \angle K + \angle (j\omega + z_1) + \angle (j\omega + z_2) + \dots - \angle (j\omega + p_1) - \angle (j\omega + p_2) - \dots$ 

- If we know the magnitude and phase responses of each term, total frequency response can be obtained by algebraic sum of each term.
- The frequency response can be simplified by utilizing straight-line approximations.
- Therefore, total frequency response can be obtained by graphic addition.





# 5.3 Bode Plot – overall plotting





## **Bode Plot**

- $\succ$  Constant K : G(s)=K
- Zeros at origin: G(s)=s
- > Poles at origin: G(s)=1/s
- Zeros at real-axis (s-plane): G(s)=(s+a)
- > Poles at real axis (s-plane): G(s) = 1/(s+b)





# Bode Plots - Constant K

• G(s) = K

 $G(s) = K; G(j\omega) = K$  $|Gj\omega|_{dB} = 20\log K$  $\angle G(j\omega) = 0^{0}$ 



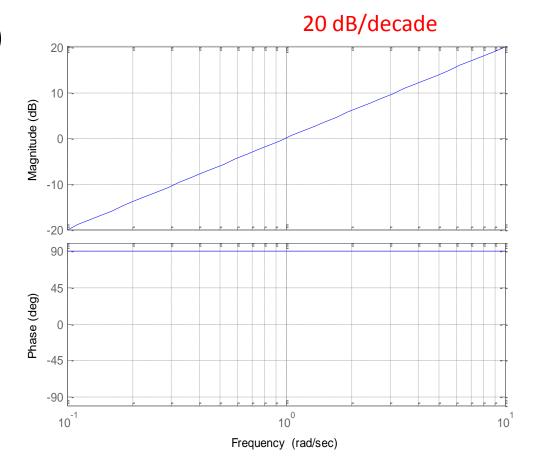


### **Bode Plots - Zeros at origin**

• G(s) = s (Zero at origin)

 $G(s) = s; G(j\omega) = j\omega$  $|Gj\omega|_{dB} = 20\log\omega$  $\angle G(j\omega) = 90^{0}$ 

• At  $\omega$  = 1, gain = 0 dB.





### **Bode Plots- Poles at origin**

• G(s) = 1/s (pole at origin)

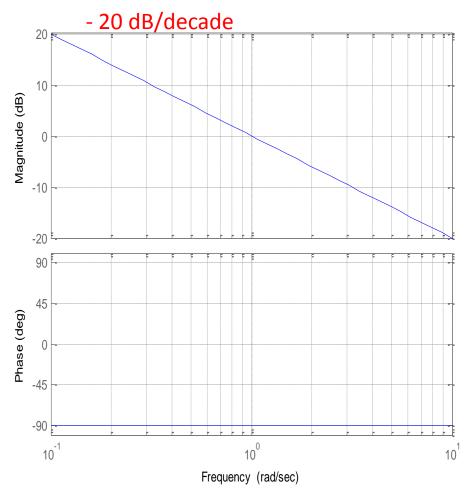
$$G(s) = 1/s; G(j\omega) = 1/j\omega$$
  

$$|Gj\omega| = \log(1/\omega)$$
  

$$|Gj\omega|_{dB} = 20\log(1/\omega) = -20\log\omega$$
  

$$\angle G(j\omega) = -90^{0}$$

• At  $\omega = 1$ , gain = 0 dB.





### Bode Plots – Zeros at real axis

• For

$G(s) = \frac{s}{-1} + 1$	
a	

• At 
$$\omega \ll a$$
,  
 $|Gj\omega|_{dB} = 20\log 1 = 0dB$   
 $\angle G(j\omega) = \tan^{-1} 0 = 0^0$ 

• At 
$$\omega = a$$
,  
 $|Gj\omega|_{dB} = 20\log\sqrt{2} = 3.01dB$   
 $\angle G(j\omega) = \tan^{-1}1 = 45^{\circ}$ 

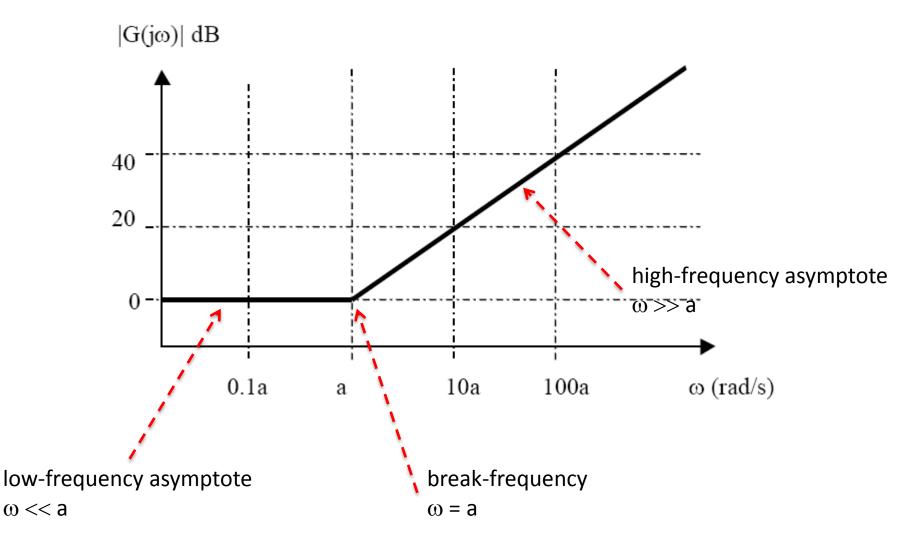
• At 
$$\omega >> a$$
,  
 $|Gj\omega|_{dB} = 20\log \omega$  20 dB/decade  
 $\angle G(j\omega) = \tan^{-1} \omega = 90^{\circ}$ 

$$G(s) = (\frac{s}{a} + 1); G(j\omega) = \frac{j\omega}{a} + 1$$
$$|Gj\omega| = \sqrt{\left(\frac{\omega}{a}\right)^2 + 1}$$
$$\angle G(j\omega) = \tan^{-1}\left(\frac{\omega}{a}\right)$$

- The low-frequency approximation is called the low-frequency asymptote.
- The frequency, ω = a is known as break frequency because it is the break between the lowand high-frequency asymptotes.
- The high-frequency approximation is called the high-frequency asymptote.



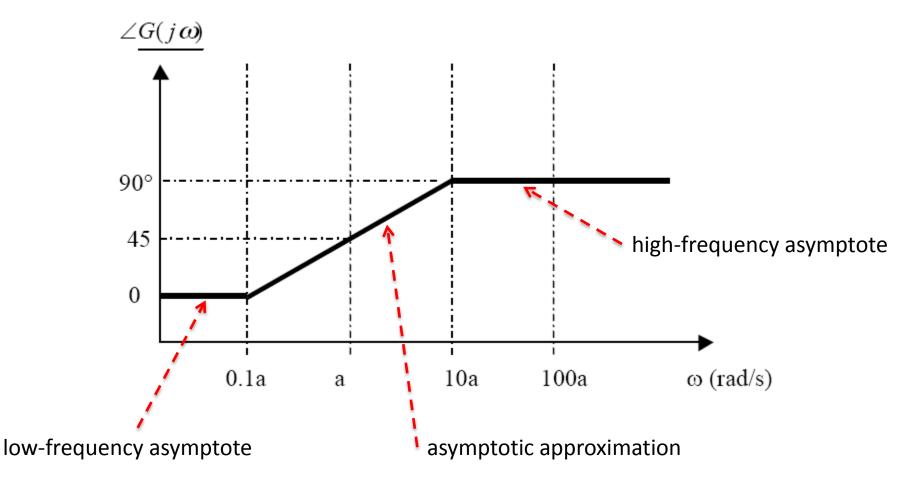
## Bode Plot- magnitude plot (Zeros)







## **Bode Plot- Phase plot (Zeros)**





Bode Plots – Poles at real axis

• For 
$$G(s) = \frac{1}{\frac{s}{a} + 1}$$
  $G(s) = \left(\frac{1}{\frac{s}{a} + 1}\right); G(j\omega) = \left(\frac{1}{\frac{j\omega}{a} + 1}\right)$   
 $|Gj\omega| = \frac{1}{\sqrt{\left(\frac{\omega}{a}\right)^2 + 1}}; \angle G(j\omega) = -\tan^{-1}\left(\frac{\omega}{a}\right)$ 

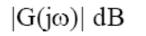
• At 
$$\omega \ll a$$
,  
 $|Gj\omega\rangle|_{dB} = 20\log 1 = 0dB$   
 $\angle G(j\omega) = \tan^{-1}0 = 0^0$ 

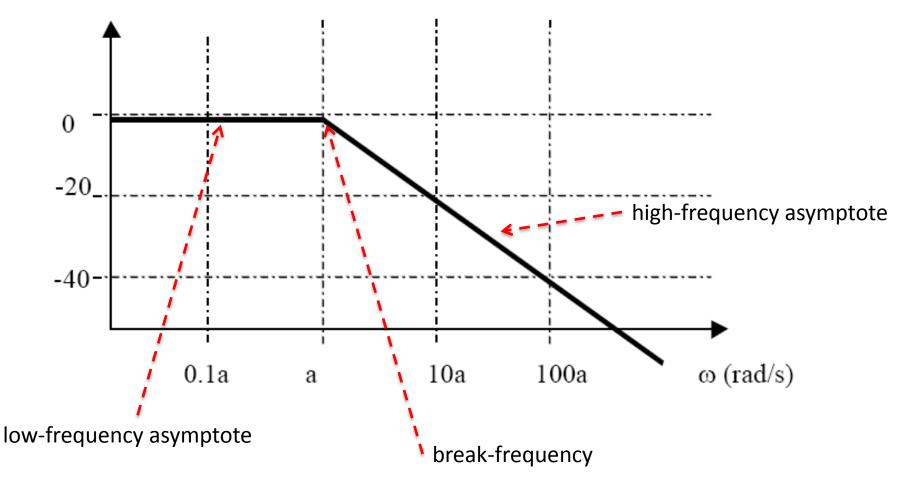
• At 
$$\omega = a$$
,  
 $|Gj\omega\rangle|_{dB} = 20\log(1/\sqrt{2}) = -3.01dB$   
 $\angle G(j\omega) = -\tan^{-1}1 = -45^{0}$   
• At  $\omega >> a$ ,  
 $|Gj\omega\rangle|_{dB} = 20\log\frac{1}{\omega}$  - 20 dB/decade  
 $\angle G(j\omega) = -\tan^{-1}\omega = -90^{0}$ 





**Bode Plots- Poles** 

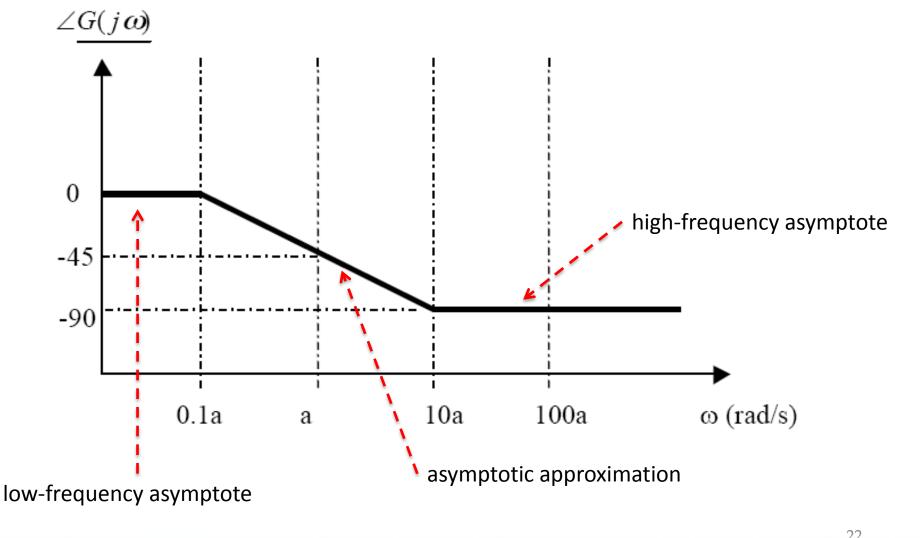








**Bode Plots- Poles** 



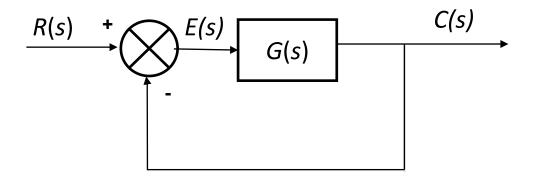




### **Exercise** 1

• Sketch the Bode plots for the system shown where

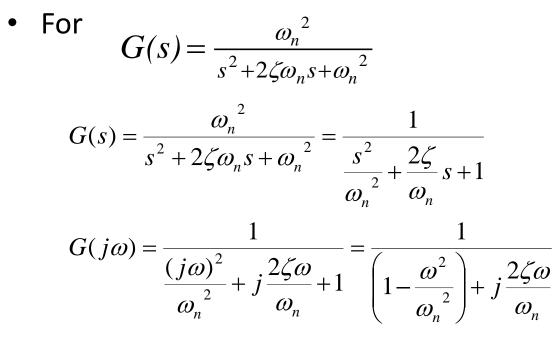
$$G(s) = \frac{K(s+3)}{s(s+1)(s+2)}$$



 Use the command in MATLAB to get the actual bode plot. (use the command 'bode(G)')



### **Bode Plots: Second Order**



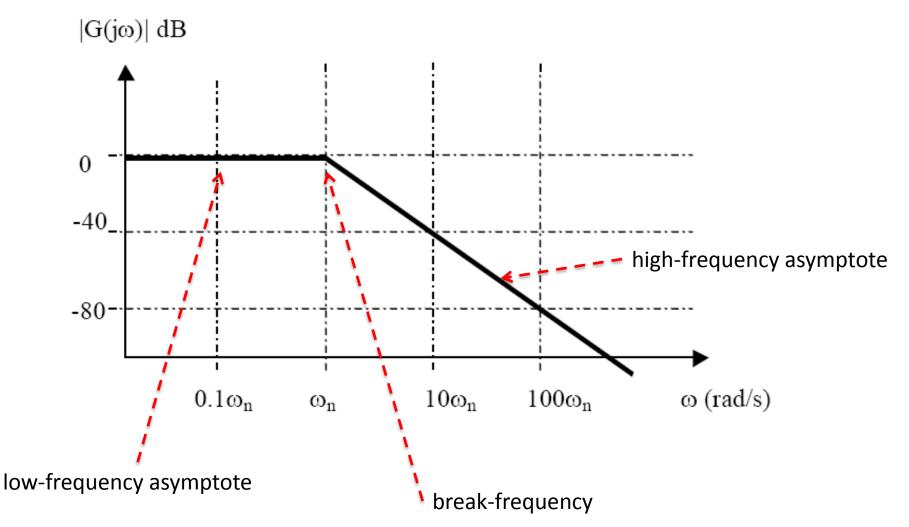
- At low frequency,  $\omega \ll \omega_{n'}$   $G(j\omega) = 1$  $|Gj\omega|_{dB} = 20\log 1 = 0 dB; \angle G(j\omega) = 0^0$
- At high frequency,  $\omega >> \omega_{n'}$ ,  $G(j\omega) = 1/-\omega^2$

$$|Gj\omega|_{dB} = -20\log\omega^2 = -40\log\omega; \angle G(j\omega) = -180^\circ$$





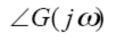
#### **Bode Plots: Second Order**

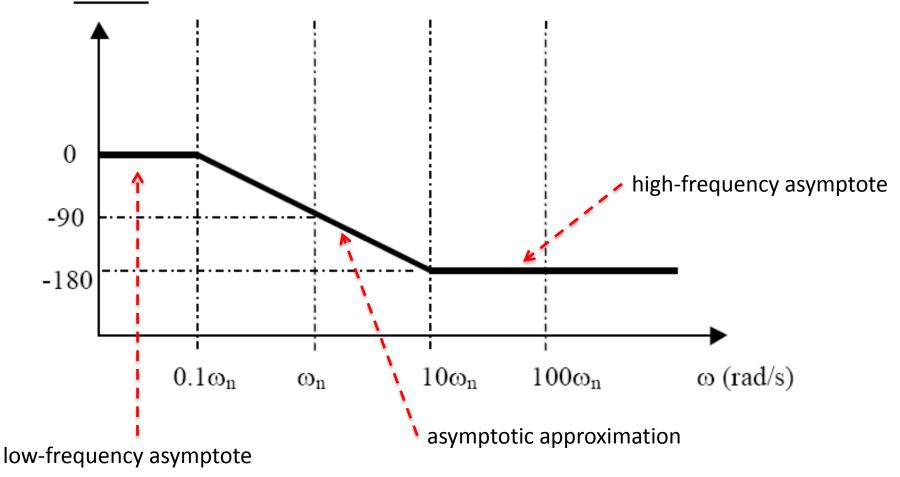






#### **Bode Plots: Second Order**







### **Bode Plots: Second Order**

$$G(s) = \frac{1}{s^2 + 2\zeta \omega_n s + \omega_n^2}$$

normalized

$$G(s) = \frac{1}{\frac{s^2}{W_n^2} + \frac{2Z}{W_n}s + 1}$$

$$G(jW) = \frac{1}{\frac{s^2}{W_n^2} + \frac{2Z}{W_n}s + 1} \bigg|_{s \to jW}$$

At the break frequency,  $\omega = \omega_n$ 

$$G(j\omega) = 1/j2\zeta$$

$$|G(j\omega)|_{dB} = -20\log(2\zeta); \angle G(j\omega) = -90^{\circ}$$





## **Example 2**

• Sketch the Bode plot for G(s) for the unity feedback system shown below where

$$G(s) = \frac{(s+3)}{(s+2)(s^2+2s+25)}$$

$$\xrightarrow{R(s)} \xrightarrow{F(s)} \xrightarrow{G(s)} \xrightarrow{C(s)}$$

Obtain the actual bode plot using MATLAB.





# 5.4 Analysis of Bode Plot





## Stability via Bode Plots

- Stability of a closed-loop system can be determined using Bode plot of an open-loop system.
- A closed-loop system is stable if the magnitude of the OL system is less than 0 dB (unity gain) at the frequency where the phase is ±180<sup>0</sup>.
- Example :
- Use Matlab to get the actual plotting
- Determine the range of K within which the unity feedback system is stable. Let G(s) = K/[(s+2)(s+4)(s+5)].
- Solution: Normalise

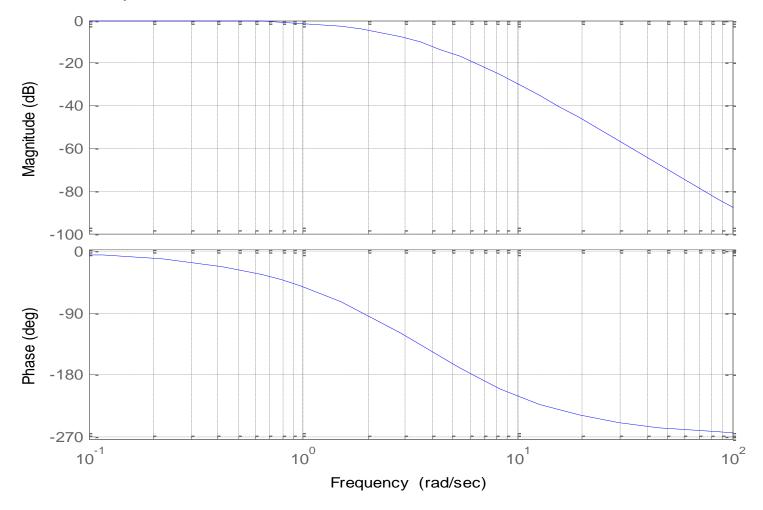
$$G(s) = \frac{K}{(40)\left(\frac{s}{2}+1\right)\left(\frac{s}{4}+1\right)\left(\frac{s}{5}+1\right)}$$

• For convenience, choose *K* = 40.





The bode plot when K = 40







## Solution 3

- With K = 40, at phase -180<sup>0</sup>,  $\omega = 7$  rad/s, magnitude = -20 dB.
- The system is stable at *K* = 40.
- Therefore, an increase in gain of 20 dB (20 log 10) is possible for stability.
- Hence the gain for stability is 400 (40 x 10).
- Range for stability: 0 < K < 400.
- Actual results:  $\omega = 6.16 \text{ rad/s}$ , K = 378.





## Gain and Phase Margins

- Gain margin and phase margin are two quantitative measures of how stable a system is.
- Systems with greater gain and phase margins can withstand greater changes in system parameters before becoming unstable.
- Gain margin,  $G_M$  is the change required in open-loop gain at 180° of phase shift to make the closed-loop system unstable.
- The gain margin is found by using the phase plot to find the gain margin frequency,  $\omega_{GM}$  where the phase angle is -180°.
- At this frequency look at the magnitude plot to determine the gain margin which is the gain required to raise or decrease the magnitude curve to 0 dB.



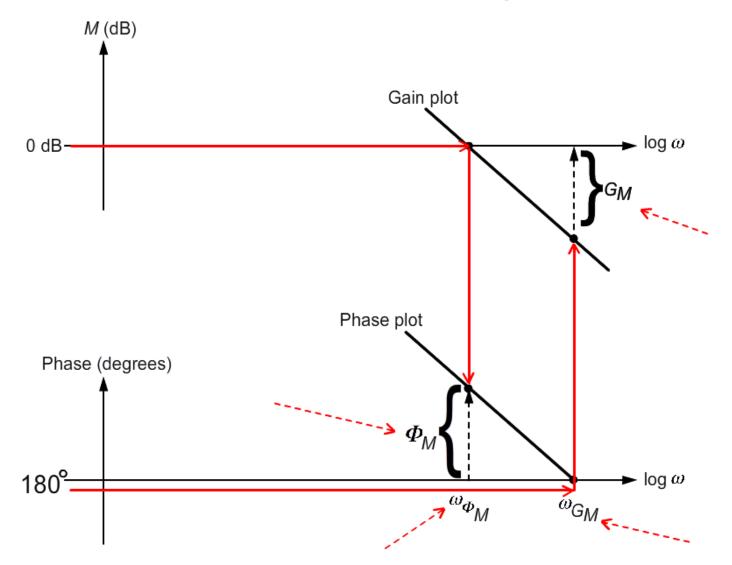


- Phase margin,  $\Phi_M$  is the change required in open-loop phase shift to make the closed-loop system unstable.
- The phase margin is found by using the magnitude curve to find the phase margin frequency,  $\omega_{\Phi M}$  where the gain is 0 dB.
- At this frequency, the phase margin is the difference between the phase value and -180<sup>0</sup>.



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**Gain and Phase Margins** 







# Example

Consider a unity feedback system :

G(s) = 200/[(s+2)(s+4)(s+5)]

Using Matlab, find the gain margin and the phase margin from the bode plot.

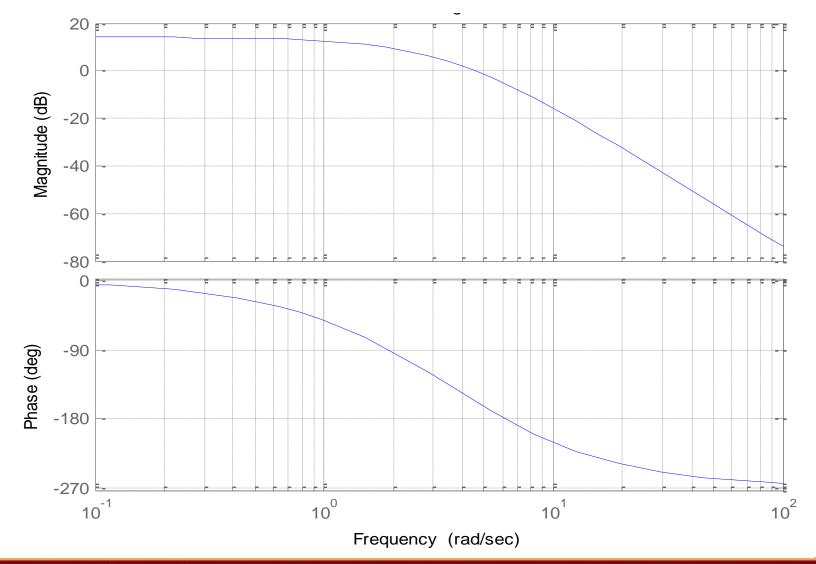
#### **SOLUTION:**

- $\omega_{GM} = 7 \text{ rad/s. } G_M = 6.02 \text{ dB.}$
- $\omega_{\Phi M}$  = 5.5 rad/s.  $\Phi_{M}$  is = 180<sup>o</sup> 165<sup>o</sup> = 15<sup>o</sup>.
- Note that: any additional zeros and/or poles will change the original bode plot. This can be clearly observed by using the command 'sisotool(G)' and adding zeros/poles from the menu. Resulted in the changes in G<sub>M</sub> and Φ<sub>M</sub>, hence the stability





#### Bode plot of the system :



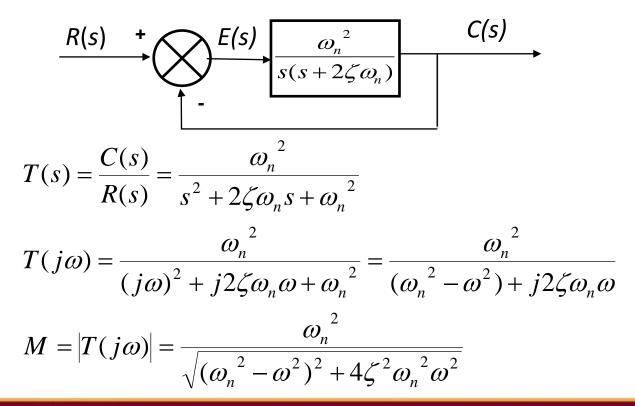
37





## Relation between Closed-Loop Time and Closed-Loop Frequency Responses

- There is a relationship between closed-loop time and closed-loop frequency responses.
- Consider the second order feedback control system:







• By plotting the closed loop bode plot and looking at the magnitude-log plot, we can measure:

- The maximum magnitude value of  $M_{p}$ ,

$$M_P = \frac{1}{2\zeta\sqrt{1-\zeta^2}}$$

- The frequency at the  $M_{
ho}$  ,

$$\omega_P = \omega_n \sqrt{1 - 2\zeta^2}$$

• Therefore, we can deduce that the maximum magnitude is directly related to damping ratio and overshoot of a system.





- Bandwidth,  $\omega_{BW}$  is the frequency at which the magnitude response curve is -3 dB.
- Relationships between the bandwidth and the time response specifications:

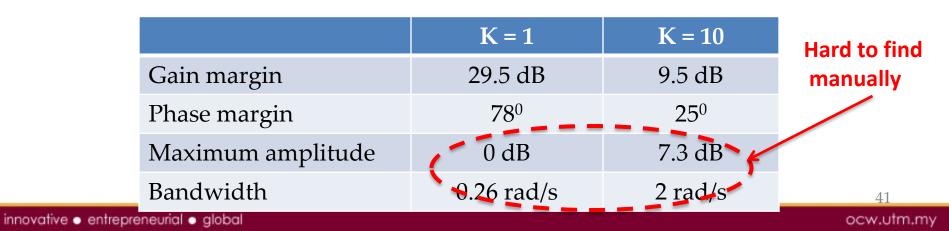
$$\omega_{BW} = \omega_n \sqrt{(1 - 2\zeta^2) + \sqrt{4\zeta^4 - 4\zeta^2}}$$
$$\omega_{BW} = \frac{4}{T_s \zeta} \sqrt{(1 - 2\zeta^2) + \sqrt{4\zeta^4 - 4\zeta^2}}$$
$$\omega_{BW} = \frac{\pi}{T_p \sqrt{1 - \zeta^2}} \sqrt{(1 - 2\zeta^2) + \sqrt{4\zeta^4 - 4\zeta^2}}$$





## **Exercise 3**

- Given a unity feedback system with G(s) = K/[s(s+1)(s+2)].
   Using Matlab, find the system gain and phase margins, maximum amplitude and bandwidth for K = 1 and 10.
   ANSWER
  - Gain and phase margins are obtained from the open-loop Bode plot. [prove using MATLAB]
  - Maximum amplitude and bandwidth can only be obtained with the closed-loop Bode plot. [prove using MATLAB]







## Relation between Closed- and Open-Loop Frequency Responses

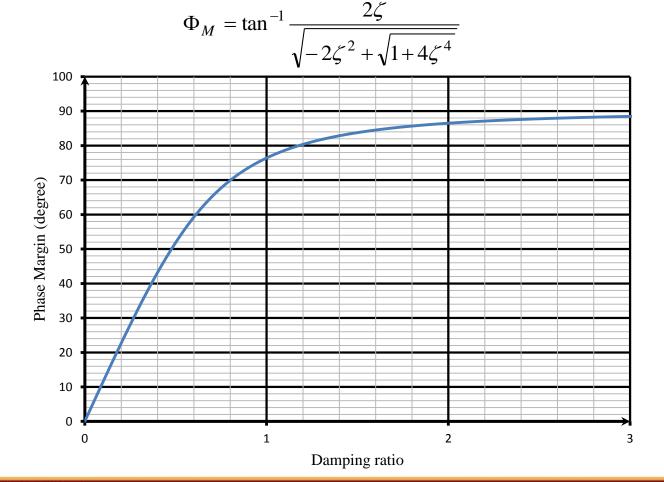
- We do not have an easy way of finding the closed-loop frequency response from which we could determine M<sub>p</sub> and thus the transient response.
- We can sketch the open-loop frequency response (Bode plot) but not the closed-loop frequency response.
- One of the techniques to obtain the closed-loop frequency response from open-loop frequency response is Nichols Chart which is not covered in this module.



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## **Damping Ratio and Phase Margin**

• The relationship between the phase margin and the damping ratio can be derived and given by:



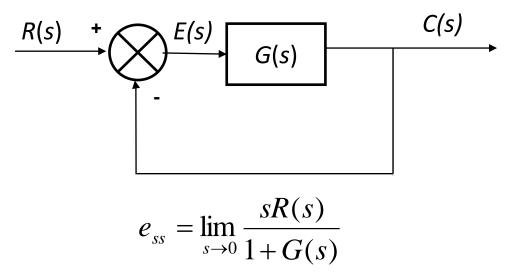
12





### Steady-state Error from Frequency Response

- The steady state error can also be found from the open loop bode plot and using the same formula from the time domain analysis
- For a unity feedback system, the steady state error can be further simplified.







- Hence, the static error constants are related to the input test signal.
- For a unit step input, the steady state error is given by,

$$e_{ss} = \lim_{s \to 0} \frac{s}{1 + G(s)} \cdot \frac{1}{s}$$
$$= \frac{1}{1 + \lim_{s \to 0} G(s)}$$
$$= \frac{1}{1 + K_P}$$

• where,  $K_P$  is the position error constant, given by

$$K_P = \lim_{s \to 0} G(s)$$





• For a unit ramp input, the steady state error is given by,

$$e_{ss} = \lim_{s \to 0} \frac{s}{1 + G(s)} \cdot \frac{1}{s^2}$$
$$= \lim_{s \to 0} \frac{1}{s + sG(s)}$$
$$= \frac{1}{\lim_{s \to 0} sG(s)}$$
$$= \frac{1}{K_{\nu}}$$

• where,  $K_v$  is the velocity error constant, given by

$$K_{\upsilon} = \lim_{s \to 0} sG(s)$$





• For a parabola input, the steady state error is given by,

$$e_{ss} = \lim_{s \to 0} \frac{s}{1 + G(s)} \cdot \frac{1}{s^3}$$
$$= \lim_{s \to 0} \frac{1}{s^2 + s^2 G(s)}$$
$$= \frac{1}{\lim_{s \to 0} s^2 G(s)}$$
$$= \frac{1}{K_a}$$

• where,  $K_a$  is the velocity error constant, given by

$$K_a = \lim_{s \to 0} s^2 G(s)$$





### Steady-state Error from Frequency Response

• For a unity feedback system with an open-loop transfer function *G(s)*, the steady-state errors can be found identify the system type and using the respective formula:

 $\frac{1}{K_a}$ 

- for system type 0: 
$$e_{ss} = \frac{1}{1 + K_P}$$

- for system type 1:  $e_{ss} = \frac{1}{K_{tr}}$ 

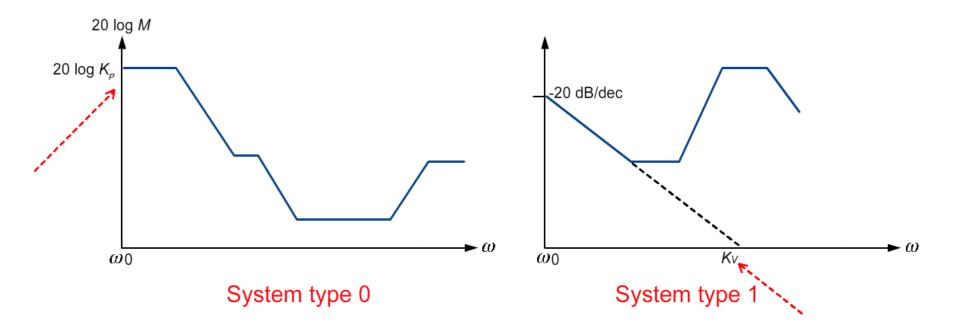
- for system type 2: 
$$e_{ss} =$$





# Steady-state Error from Frequency Response

• By identifying the system type from the open-loop Bode plot, the steady state error can be easily found as follows,

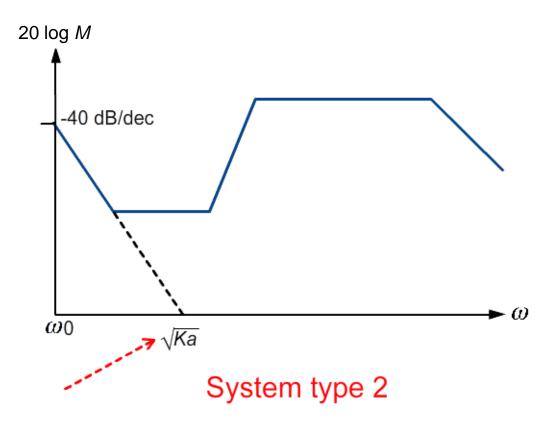


19





# Steady-state Error from Frequency Response

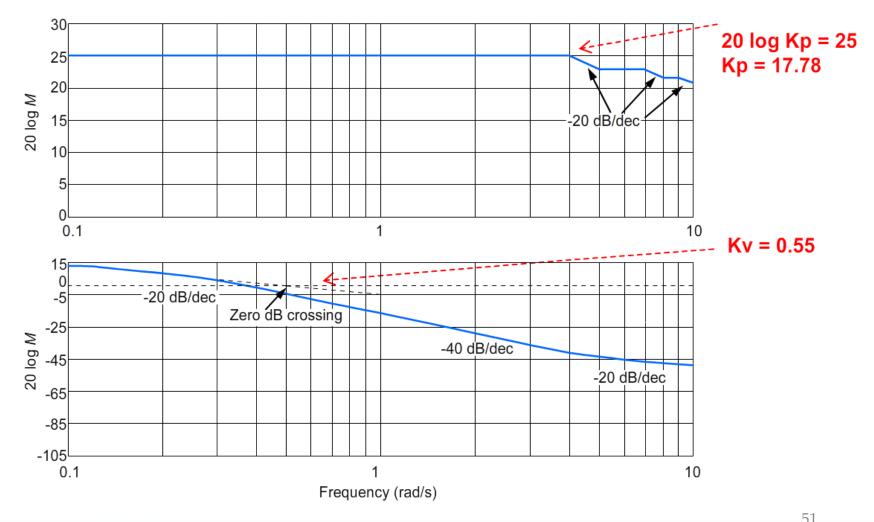




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## Example

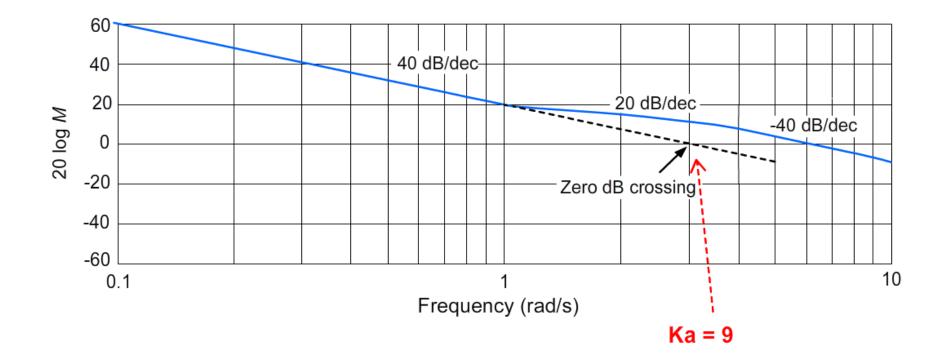
• Find the steady-state error for the following Bode plots:





### 

## Example







## Conclusions

We have covered

- ✓ The graphical analysis using Bode Plot
- ✓ The stability analysis by looking at Gain and Phase Margins
- ✓ Some relationships between open loop and closed loop systems' information





#### REFERENCES

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