

CREEP FAILURE

- The start of tertiary creep indicates that damage in the metal had occurred, which will end in creep failure.
- “But why should tertiary creep occur at all?”
- It has been shown that in steady-state creep there is a balance between two opposing tendencies: The trend for the strength of the material to increase during creep by strain hardening and the trend for the strength to decrease by recovery.
- “What happens to upset this balance when tertiary creep begins?”



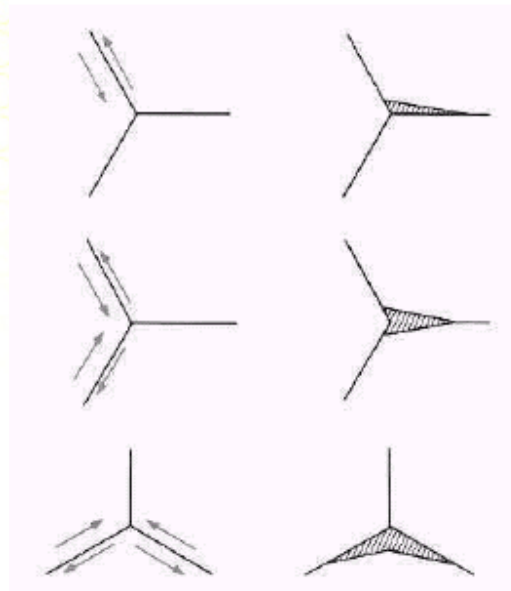
CREEP FAILURE

- Steady-state creep is only a metastable state that can be brought to an end by irreversible changes
- The onset of creep is an indication that **voids or cracks** are forming in the material, the number of these voids increases with strain and time. By reducing the net cross-sectional area of load bearing material, these voids must weaken the material and help to induce tertiary creep.
- Two types of voids have been mainly observed in alloys after creep: **round** and **wedge shaped** voids
- The mechanism of void formation involves **grain boundary sliding**



- Wedge crack:

- ✓ Initiate mostly at grain boundaries which are aligned for max shear.

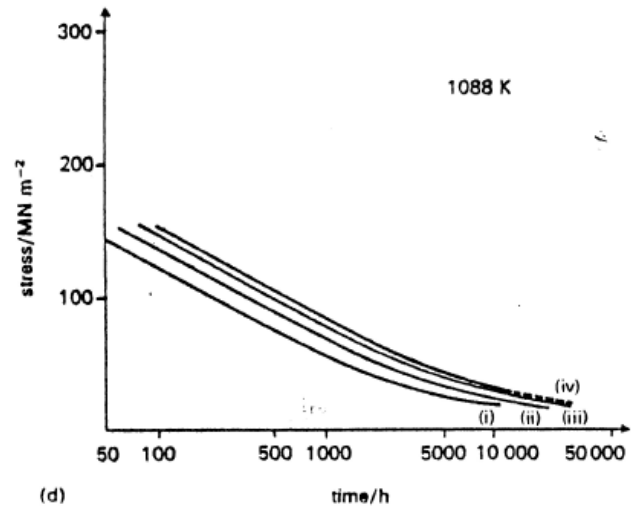
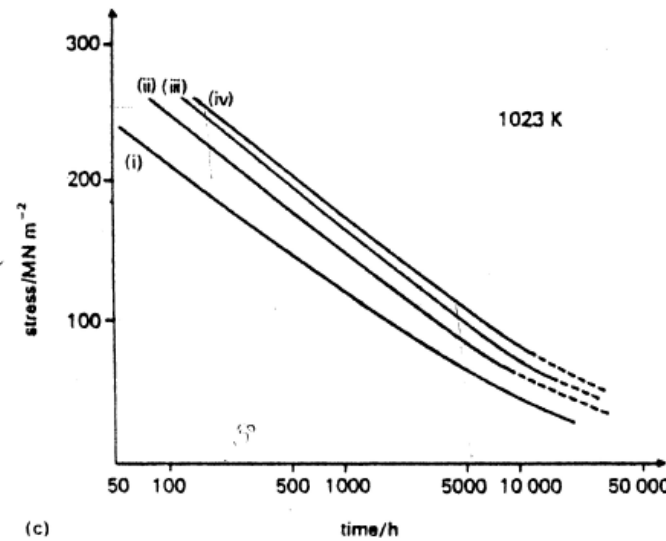
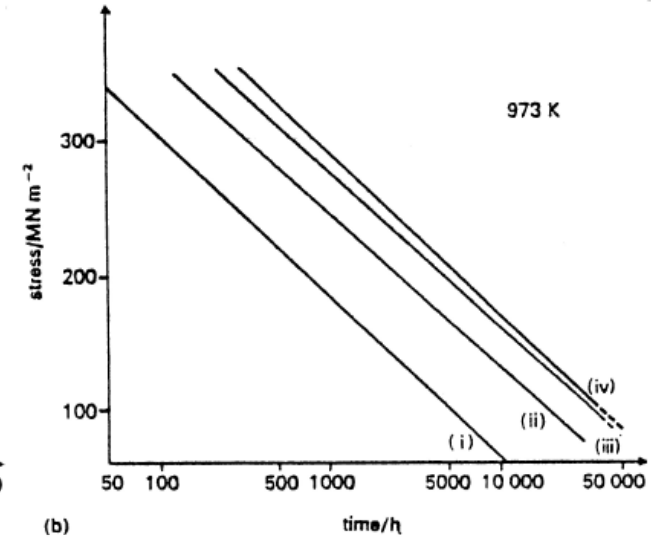
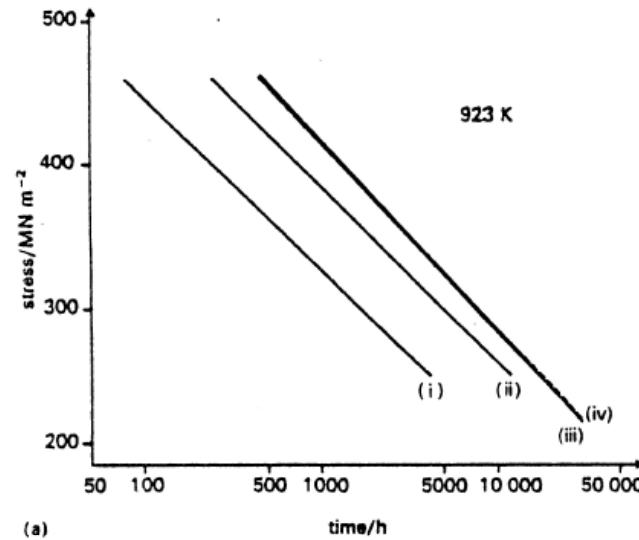


- Round or elliptical cavities (r-cracks)
 - ✓ form in the grain boundaries that are aligned normal to the tensile stress

Presenting Creep Data

- Creep deformation involves 4 major variables: stress, strain, time and temperature
- The method of presenting data used depends on the particular question to which a design engineer requires an answer
- One important information is **the time it will take** a specimen of a material to reach a **particular creep strain at a specified temperature**
- This is provided by plotting *isometric stress-time* curves

Strain
(iv) > (iii) > (ii) > (i)

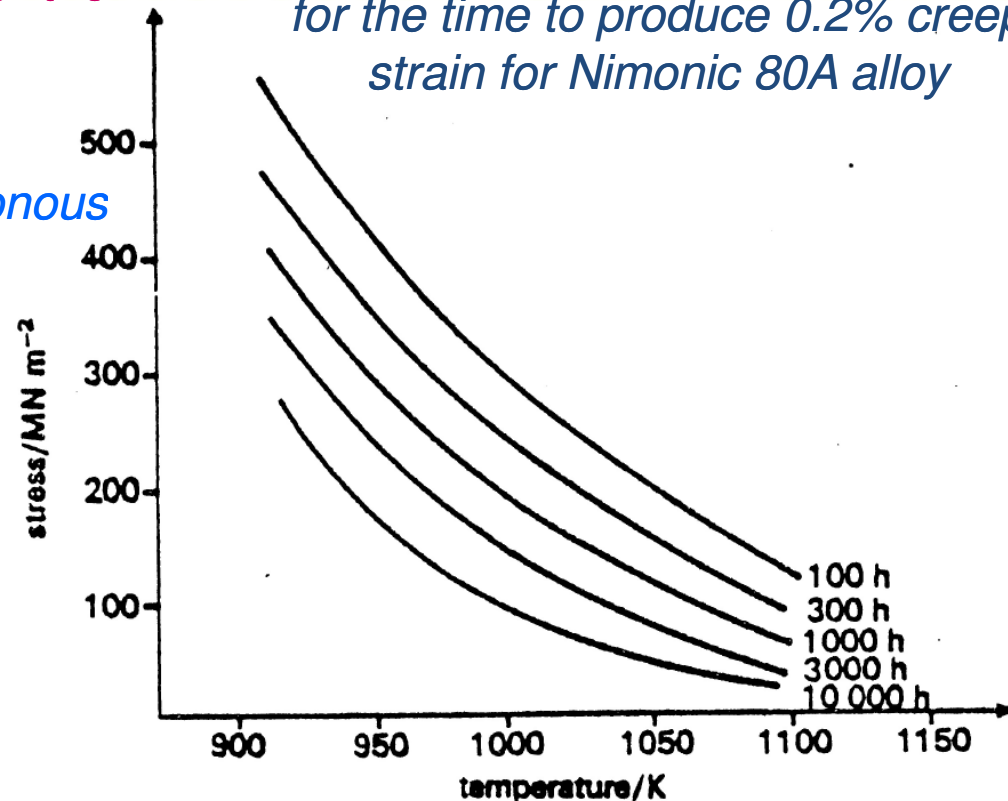


Isometric stress-time curves for different strains



- Other important question an engineer might ask is:
- My component must not exceed a creep strain of 0.1% in 500 hours, **What is the maximum stress the material will support at 1000 K?**
- In this case we use *isochronous stress-temperature curves*

Isochronous stress-temperature curves for the time to produce 0.2% creep strain for Nimonic 80A alloy



Prediction of long-time properties

- For long-time creep and stress rupture data
 - ✓ 1% deformation in 100,000 h (11.4 years!!!)
 - ✓ Impractical to collect data from normal laboratory test.
 - ✓ Therefore, we need to perform creep test/creep rupture test at temperatures in excess, and making suitable extrapolation to the in-service condition.



Larson-Miller Parameter

- Stress rupture or failure data for high temperature resistant alloys are often plotted as *log stress to rupture vs. a combination of log time to rupture and temperature*. The *Larson-Miller* (LM) parameter is most widely used.

$$P(\text{Larson-Miller})\text{parameter} = T(\log t_r + C)$$

- T , is the temperature in K.
- t_r , stress-rupture time, hours
- C , constant, is assumed to have a value of 20.

- In terms of K-hours:

$$P(LM) = \{T(0C) + 273(20 + \log tr)\}$$

- According to the *Larson-Miller* parameter, at a given stress level the log time to stress rupture (failure) plus a constant multiplied by the temperature remains constant for a given material.



CREEP CONTROL

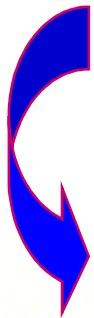
- The objectives of creep control are to produce (use) a material that is stable under specified levels of stress, temperature and environment.
- The material should not change its dimensions (or creep) and should not lose its integrity or fracture.
- In practice true stability at high temperature has never been achieved and the development of alloys is a matter of trying to retard inevitable changes in the material
- “There is a finite lifetime for a component in service under stress at high temperature”



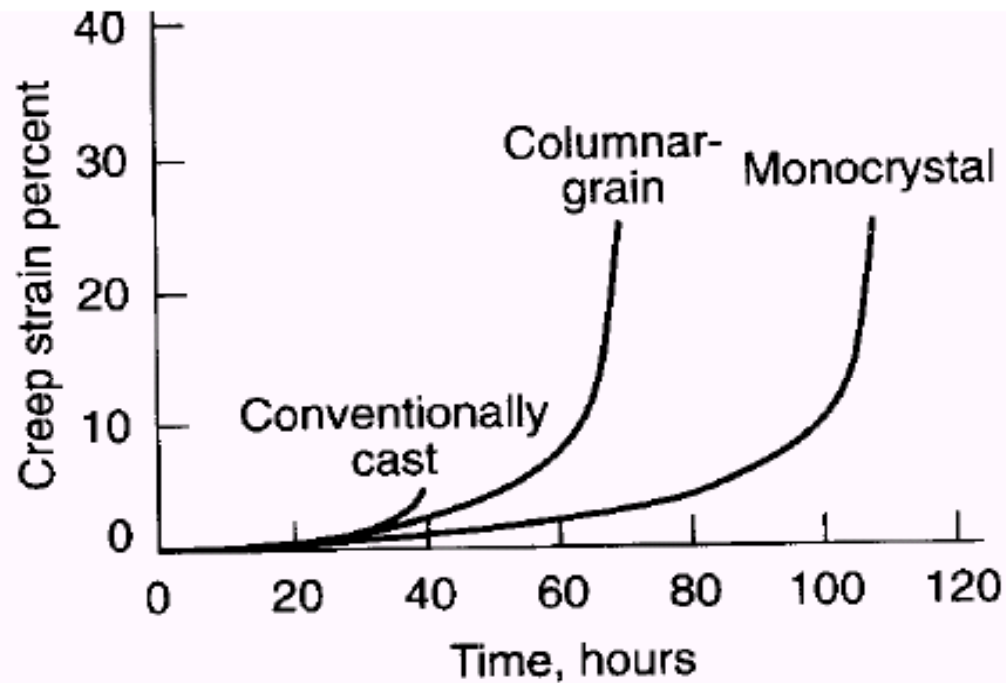
Example: turbine blades for jet engines

- The strategy used to develop Nickel-based superalloys for turbine blades has 3 aims:

1. to inhibit the oxidation of the alloy (add Chromium)
2. to inhibit the deformation of the grains (add Aluminium and titanium)
3. to inhibit the deformation between the grains.



- Introduce precipitates at the grain boundaries (carbides) which reduce grain boundary sliding
- Align the grain parallel to the applied stress (directional solidification)
- Completely remove the grain boundaries (single crystal turbines)



Comparison of creep properties at 980°C and 207 MPa of MAR-M200 in equiaxed casting, DS and SC turbine blades



References:

- **Callister W.D., Materials Science and Engineering – An introduction, 7th edition, Wiley, 2007.**
- **Smith W.F., Foundation of Materials Science and Engineering, 4th edition, McGraw Hill, 2006.**
- **Fontana M.G., Corrosion Engineering, 3rd edition, McGraw Hill, 1991.**
- **Dieter G.E., Mechanical Metallurgy, 3rd edition, 1991.**

